

LOCKHEED AIRCRAFT CORP.		ENGINEERING STUDY <input type="checkbox"/>		CHANGE PROPOSAL <input checked="" type="checkbox"/>		LAC-98				
DATE 4-3-61		AFFECTS:		WSPO <input checked="" type="checkbox"/>		PROJECT <input checked="" type="checkbox"/>				
NAME OF MAJOR COMPONENT AIRPLANE		PART OR LOWEST SUBASSEMBLY FUEL PUMP		PART NO. & MODEL OR TYPE						
TITLE OF PROPOSAL : FUEL SYSTEM REVISION										
NATURE OF PROPOSAL : SEE PAGE 2										
REASON FOR PROPOSAL : 1. To improve reliability of the existing hydraulically driven fuel boost pump by replacing the present chip catcher with a 200 mesh strainer. 2. To reduce the possibility of engine flameout due to loss of fuel pressure particularly at high altitudes by providing an electrically driven fuel boost pump as backup for the existing hyd. driven pump. The new pump will be controlled by the pilot and is to be turned on for all operations above 50M ft.										
ES		ESTIMATED COST AND TIME INVOLVED : - - - ADDITIONAL FUNDING REQUIRED : - - -								
CP		ESTIMATED COST FOR KITS OR PARTS : See Pages 3 & 4 ADDITIONAL FUNDING REQUIRED : None								
ITEMS AFFECTED BY PROPOSAL :										
SAFETY <input type="checkbox"/>	MISSION EFFEC- TIVENESS <input checked="" type="checkbox"/>	PERFORM- ANCE <input checked="" type="checkbox"/>	OPERATING PROCEDURE <input checked="" type="checkbox"/>	INTER- CHANGE- ABILITY <input type="checkbox"/>	WEIGHT OR WEIGHT & BALANCE <input checked="" type="checkbox"/>	TOOLS & SUPPORT EQUIPMENT <input type="checkbox"/>	MAINTENANCE PROCEDURE <input checked="" type="checkbox"/>	SERVICE LIFE <input type="checkbox"/>	FLIGHT MANUAL <input checked="" type="checkbox"/>	MAINTENANCE MANUAL <input checked="" type="checkbox"/>
EST. MAN/HRS. REQ'D. TO ACCOMPLISH CHANGE IN FIELD										
SOURCE OF PARTS FOR KIT LAC				AVAILABILITY - - WEEKS AFTER APPROVAL See Page 4						
DISPOSITION OF SPARES AFFECTED The following will no longer be used: H-80 Chip Catcher; Revere Float Switch P/N F8300-8 (F8300-26 on U-2C a/c)								STAT		
INITIATED BY : LAC				APPROVED : WSPO				4/10/61		

NATURE OF PROPOSAL:

Design Study

1. Modify the fuel system in one aircraft (692/359) by installing an electrically driven fuel boost pump with related plumbing, wiring and cockpit controls. This has been accomplished as Contract SP-1918 Product Improvement.
2. Performance data to be obtained from operation of the aircraft at LAFB over a significant period (approximately 3 months).
3. This test installation will be replaced by the standard installation (outlined below) at a convenient time at the conclusion of operational tests.

Change Proposal

1. Modify the fuel system on all aircraft (except serials 342 & 358)* as follows:
 - a. Replace the existing Chip Catcher (P/N H-80) with a 200 mesh strainer.
 - b. Install a submerged A.C. electric motor driven boost pump in the right-hand sump tank. Install related plumbing to connect pump in parallel with the existing boost pump and bypass line. This includes the addition of two new check valves; one in each boost pump fuel out line, and a pressure switch between the check valve and the new pump (see attached diagram).
 - c. Replace the existing sump tank overflow light in the cockpit with a fuel pressure indicator light (elec. boost pump only) and install pilot's control switch. Install power relay on Q-bay "CB & Relay" panel and install system wiring.
 - d. On all aircraft remove the overflow float switch from the sump tanks.

* NOTE: These fuel system modifications previously authorized for incorporation on a/c serials 342 and 358 in conjunction with In-Flight Refueling provisions under approved ECP No. LAC-101.

2. Prepare and issue a Service Bulletin.
3. Fabricate appropriate aircraft provisioning kits.
4. Installation of kits can be accomplished in the field. Modification of sump tanks to incorporate pump flange must be done at the factory on a turn around basis unless the entire program is scheduled for IRAN.

ESTIMATED COST FOR KITS OR PARTS:

STAT

Customer No. 1

Six (6)* kits @ [] kit (SP-1917)

Modification of R/H Sump Tank (6 ea.) (SP-1918)
[] Tank

RECOMMENDED SPARES (SP-1917)

STAT

Part No.	Description	Qty.	Unit Price	Total Price
301385	Strainer Assy.	2		
8501-1	Strainer Screen	4		
** 219200	Pump	6		
310900	Check Valve	5		
M410G-10A-42	Press. Switch	10		
AV16A1185	Valve	5		
475C-58NW	Valve	5		

TOTAL PRICE Customer #1

STAT

* Two kits to be manufactured and installed under the Aerial Refueling System Program.

** Quantity factored by four (4) spare pumps ordered on P.R. 61-147 for ECP-101 Aerial Refueling System.

Customer to obtain spares for the following from AF assets:

MS 35058-22 Switch
AN 3312-1 Relay

Cost Recap - Customer #1

Total Cost SP-1917
Total Cost SP-1918

GRAND TOTAL

STAT

ESTIMATED COST FOR KITS OR PARTS:

STAT

Customer No. 2

Thirty-one (31) kits @ kit (SP-1917)

Modification of R/H Sump Tank (31 ea.) (SP-1918)
 tank

STAT

RECOMMENDED SPARES (SP-1917)

STAT

Part No.	Description	Qty.	Unit Price	Total Price
301385	Strainer Assy.	4		
8501-1	Strainer Screen	10		
219200	Pump	16		
310900	Check Valve	12		
M410G-10A-42	Press. Switch	24		
AV16A1185	Valve	12		
475C-58NW	Valve	12		

TOTAL PRICE Customer #2

Customer to obtain spares for the following from AF assets:

STAT

MS 35058-22 Switch
AN 3312-1 Relay

Cost Recap - Customer #2

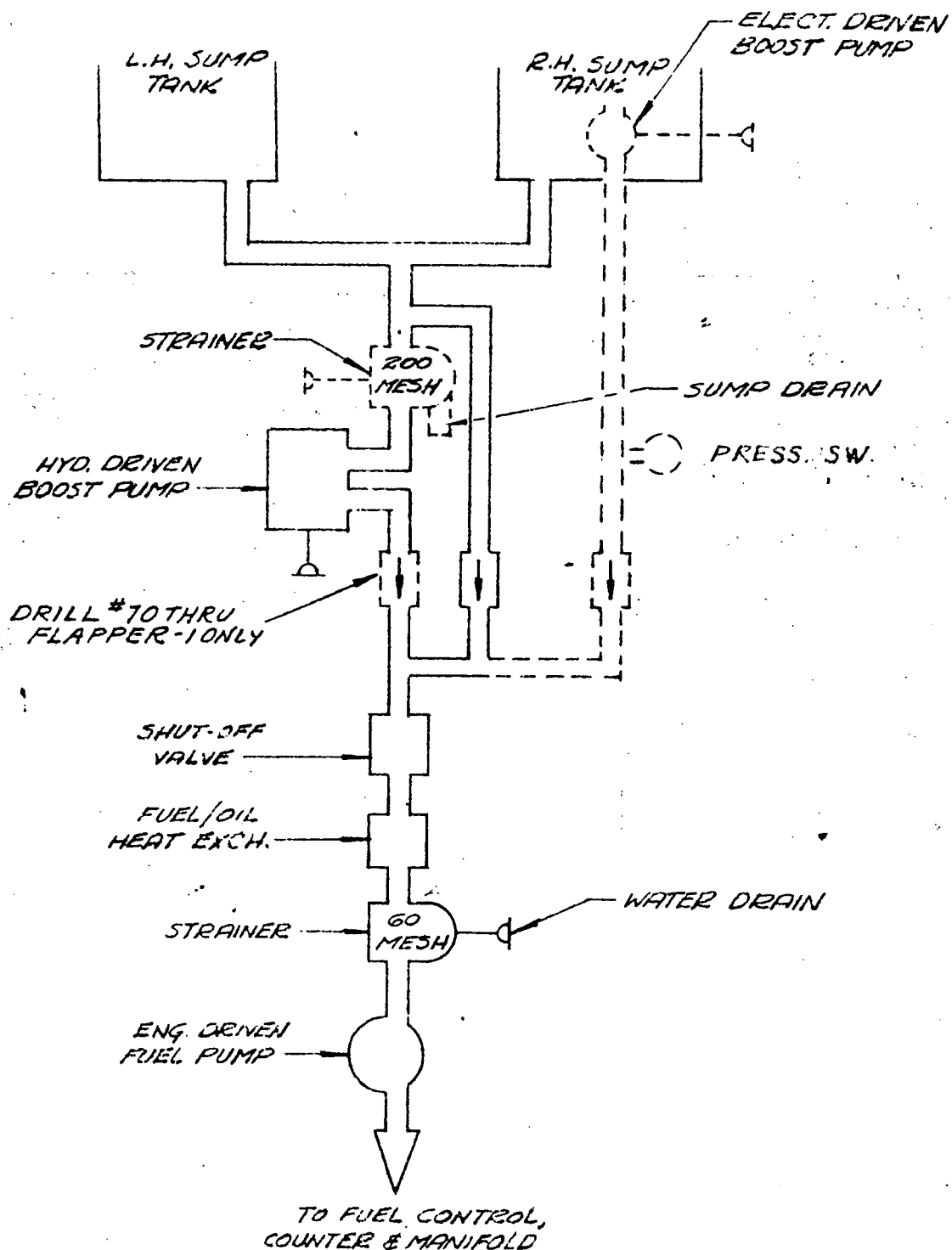
Total Cost SP-1917
Total Cost SP-1918

GRAND TOTAL

STAT

SCHEDULE (Both Customers)

180 days from date of approval



FUEL SYSTEM DIAGRAM

SHOWING ADDED ELECTRICALLY DRIVEN
 SUBMERGED BOOST PUMP & 200 MESH STRAINER

- EXISTING SYSTEM
- - - ADDED COMPONENTS